

ELECTRICAL AND MECHANICAL PROPERTIES OF COMPOSITE MATERIALS BASED ON CARBON NANOTUBES FOR AEROSPACE APPLICATIONS

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Synthesis and Characterization



Introduction

Several methods have been used to produce single-walled as well as multi-walled carbon nanotubes. The diameter of the MWNTs typically range from 10 to 50 nm, while the length exceeds 10 micron. For SWNTs, the diameter is ~ 1 nm and the lengths can be up to 100 microns. CNTs exhibit unique physical and chemical properties. Mechanically, they have high structural perfection in their structures and the highest modulus of all known materials. Owing to their extreme properties, lightweight and chemical inertness, CNTs yield suitable candidates for aerospace applications, such as CNT-based electron field emitters and composite materials for extreme conditions (large temperature and pressure gradients, radiation).

Synthesis of CNT

CNTs were synthesized in a dc arc plasma system in helium atmosphere at a pressure of 600 torr. Arc was struck between two high purity graphite rods which served as electrodes. The discharge is carried out at 20 V voltage and a current in the range of 100 – 150 A. Some amount of the evaporated carbon condenses on the tip of the cathode, forming a slag-like hard deposit (cathode deposit). The deposit consists of two regions, viz., an inner fibrous black core and an outer gray, hard shell. The inner core has a columnar structure that is made up of bundles of CNTs and flocks of polyhedral carbon particles. On the other hand, the hard shell is made of polycrystalline graphite. CNTs thus synthesized (without any purification treatments were used for field emission studies. Using a phosphorescence screen coated on a conducting glass the emitted electrons result in a bluish glow.

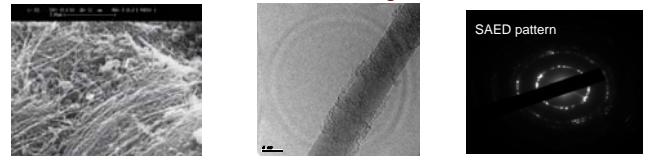
Field emission studies



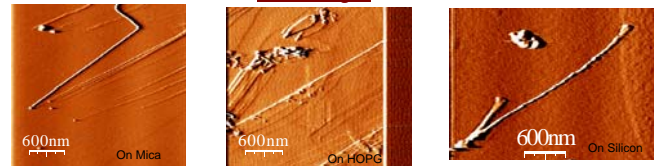
- Turn on electric field:
 - $E = 0.4 \text{ V}/\mu\text{m}$ (current density was about $10 \mu\text{A}/\text{cm}^2$)
- Current density:
 - $J \sim 10 \text{ mA}/\text{cm}^2$ (when the electric field E is $1 \text{ V}/\mu\text{m}$)
- Current stability:
 - $I = 377 \pm 17 \mu\text{A}$ (mean value of current at 1 kV over a duration of 1 hour)



SEM and TEM Images



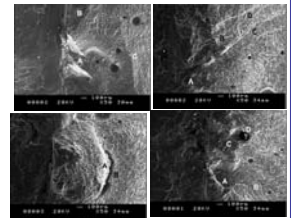
AFM Images



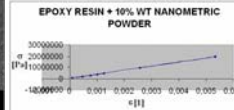
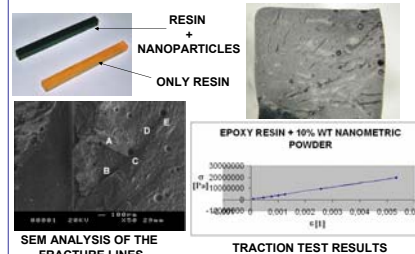
COMPOSITE WITH CARBON NANOTUBES

- RESIN, CURING AGENT, NANOPARTICLES
- SPECIMEN PREPARATION
- UNIFORM DISTRIBUTION OF THE NANOMETRIC PARTICLES (GRAPHITE, CARBON NANOTUBES, ETC.) IN THE POLYMERIC MATRIX
- MECHANICAL TEST
- MORPHOLOGICAL ANALYSIS
- COMPOSITE MECHANICAL FRACTURE BEHAVIOUR
- COMPOSITE CHARACTERIZATION
- AEROSPACE STRUCTURE APPLICATIONS
- THEORETICAL AND NUMERICAL STUDIES

COMPOSITE WITH CARBON NANOTUBES (DYNAMIC TEST ANALYSIS)

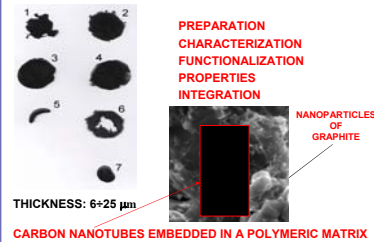


TEST OF COMPOSITE SPECIMENS



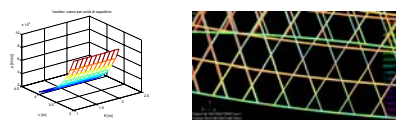
WITH NANOMETRIC PARTICLES (GRAPHITE AND CARBON NANOTUBES) THE YOUNG MODULUS IMPROVEMENT IS LARGER THAN 12%

NANOSTRUCTURED THIN FILM



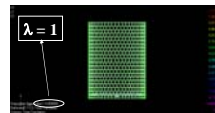
ANISOGRID LATTICE STRUCTURE DESIGN

REQUIREMENTS ANALYSIS: MINIMUM MASS – STATIC RESISTANCE – STABILITY

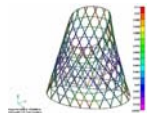


NUMERICAL MODEL

STATIC RESISTANCE ANALYSIS



BUCKLING ANALYSIS



DYNAMIC ANALYSIS

ANISOGRID LATTICE STRUCTURES PRELIMINARY FLAT PROTOTYPE REALIZATION

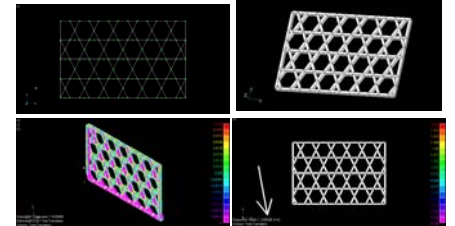


ANISOGRID LATTICE STRUCTURES FLAT AND CYLINDRICAL PROTOTYPE REALIZATION

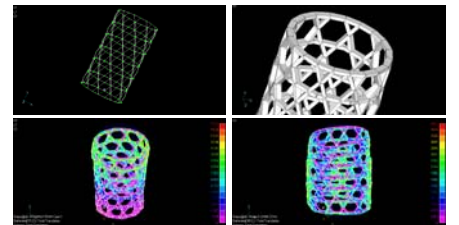


CAD MODEL POSITIVE MOULD SILICON NEGATIVE MOULD MANDREL FOR PROTOTYPE PRODUCTION USING FILAMENT WINDING

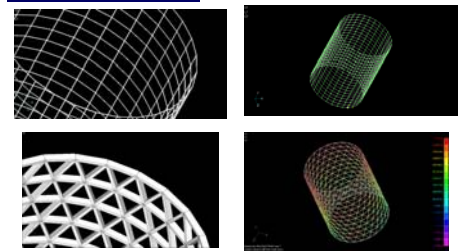
ANISOGRID LATTICE STRUCTURES FLAT PROTOTYPE FEM ANALYSIS



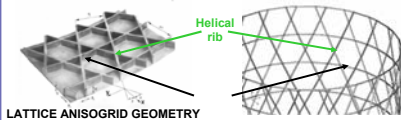
ANISOGRID LATTICE STRUCTURES CYLINDRICAL PROTOTYPE FEM ANALYSIS



ISOGRID LATTICE STRUCTURES RECTANGULAR AND TRIANGULAR CELLS



ANISOGRID LATTICE STRUCTURES FOR AEROSPACE APPLICATIONS



USING VASILEV MODEL WE GET AN ANISOGRID LATTICE STRUCTURE WITH REQUIREMENTS: MINIMUM MASS STATIC RESISTANCE LOCAL AND GLOBAL STABILITY (BUCKLING) WITH THE CARBON NANOTUBES IT IS POSSIBLE TO OBTAIN A VERY HIGH MASS REDUCTION OF THE ANISOGRID LATTICE STRUCTURE

USING, FOR EXAMPLE, THE STRUCTURE PARAMETERS BELOW: radius $R = 1.5 \text{ m}$ height $L = 4 \text{ m}$ helical ribs number $n_r = 20$ Applied load $P = 3 \text{ MN}$

VARYING THE MATERIALS USED:

MASS MATERIALS	Al (2024)*	HsHp**	5% C%***
CALCULATED MASS M [Kg]	206	84	69

* = aluminum alloy 2024

** = epoxy resin and uni-directional traditional carbon fibers composite (HsHp)

*** = the above composite HsHp with a 5% in wt of carbon nanotubes embedded in the matrix